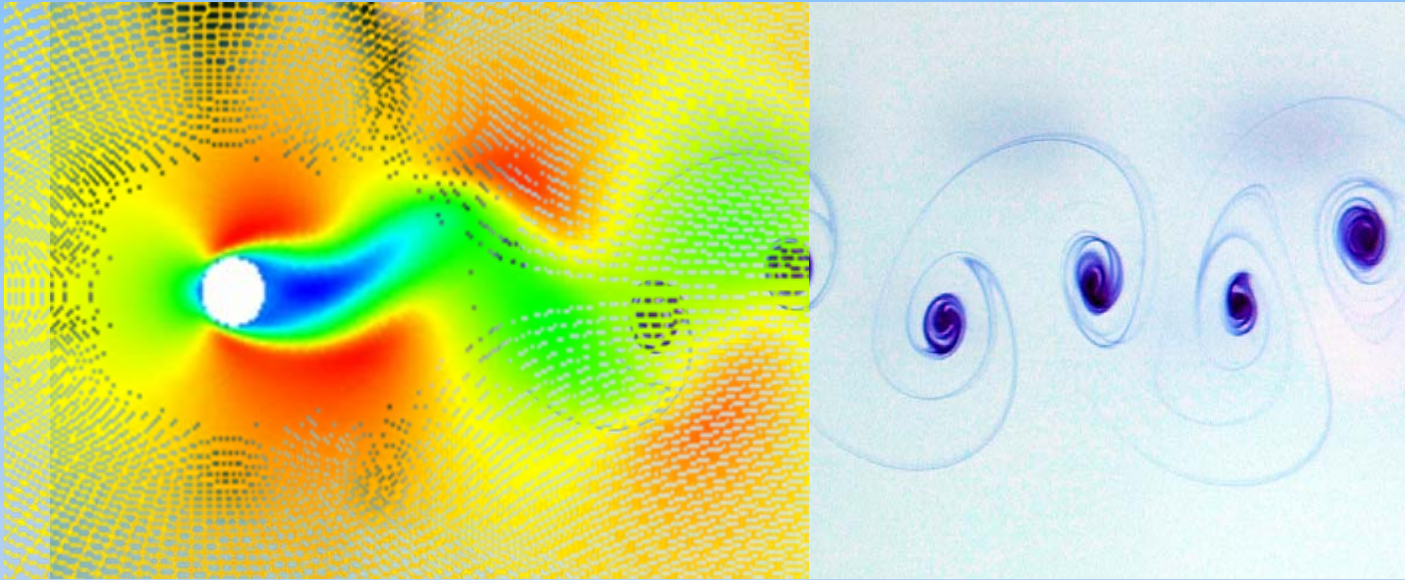




Feedback Control of a Circular Cylinder Wake



*Integrating CFD and Experiments in Aerodynamics
20-21 June, 2007*

*Jürgen Seidel
Stefan Siegel
Kelly Cohen
Thomas McLaughlin*

Report Documentation Page			Form Approved OMB No. 0704-0188		
Public reporting burden for the collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to a penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number.					
1. REPORT DATE JUN 2007		2. REPORT TYPE N/A		3. DATES COVERED -	
4. TITLE AND SUBTITLE Feedback Control of a Circular Cylinder Wake				5a. CONTRACT NUMBER	
				5b. GRANT NUMBER	
				5c. PROGRAM ELEMENT NUMBER	
6. AUTHOR(S)				5d. PROJECT NUMBER	
				5e. TASK NUMBER	
				5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) USAFA Department of Aeronautics				8. PERFORMING ORGANIZATION REPORT NUMBER	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)				10. SPONSOR/MONITOR'S ACRONYM(S)	
				11. SPONSOR/MONITOR'S REPORT NUMBER(S)	
12. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release, distribution unlimited					
13. SUPPLEMENTARY NOTES The original document contains color images.					
14. ABSTRACT					
15. SUBJECT TERMS					
16. SECURITY CLASSIFICATION OF:			17. LIMITATION OF ABSTRACT UU	18. NUMBER OF PAGES 44	19a. NAME OF RESPONSIBLE PERSON
a. REPORT unclassified	b. ABSTRACT unclassified	c. THIS PAGE unclassified			



What is Feedback Flow Control?

- Flow Control

- Influence the flow field to achieve a desired effect using *minimal* actuation power
 - Passive
 - Vortex generators
 - Active
 - Synthetic jets
 - Time-dependent (periodic) blowing and suction
 - Piezo-electric micro components

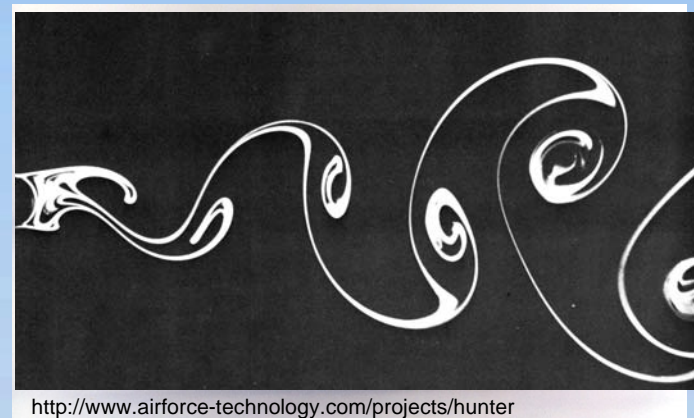
- Feedback

- Sensors in the wake measure instantaneous flow quantities (velocity, pressure) at given points
- Actuation based on sensor information



Why Feedback Flow Control?

- Bluff bodies (e.g. on UAVs) serve vital *operational functions*, but they are *aerodynamically detrimental*
- *Flow separation* results in a wake behind the bluff body
 - characterized by unsteady *vortex shedding*
 - results in drag, noise, and vibration
 - is detrimental for operation, structural integrity
- “Passive” designs are impractical or inhibit functionality
- “Active” methods are point designs
- ***Feedback flow control*** is an effective way of suppressing self-excited flow oscillations without modifying the geometry





How to Feedback Flow Control?

Modeling wake dynamics
for controller design

Model Independent
Approach

- Simple to implement experimentally
- Little success in past 30 years

Direct Navier Stokes
Approach

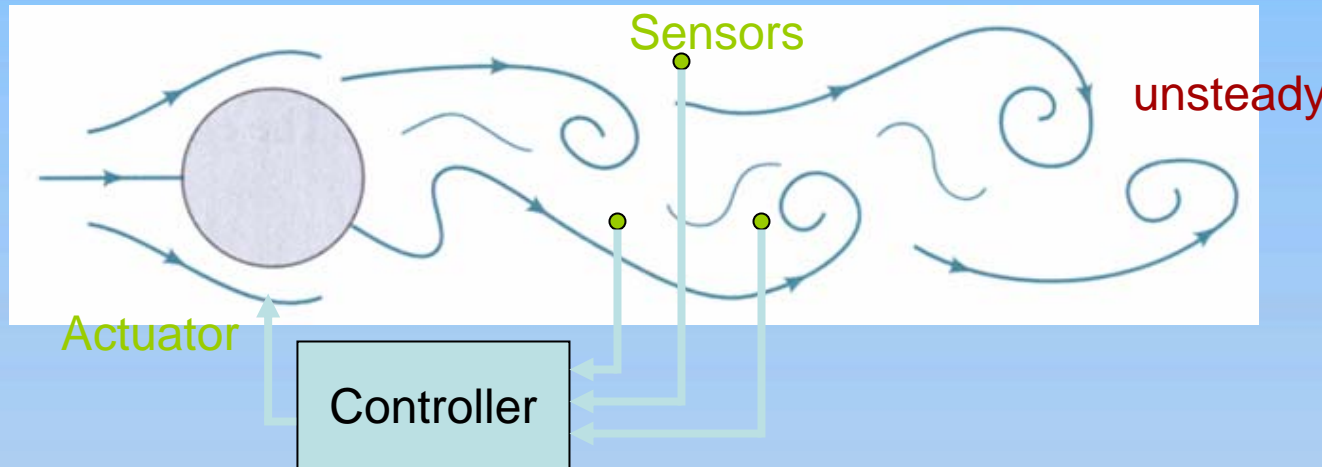
- Ideal control approach, complete set of equations
- Computationally very intensive
- Cannot be implemented in real time in the near future

Low-Dimensional
Approach

- Recent developments in effective low-dimensional models
- Can be implemented with relative ease
- Model building is tough



Cylinder Wake Feedback Control

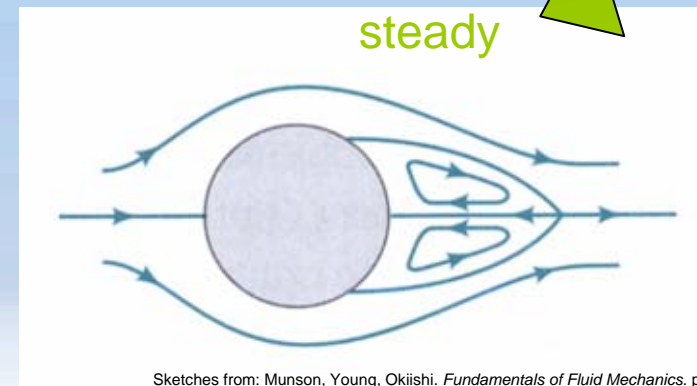


Goal: Develop a feedback control strategy to suppress the vortex street of a cylinder at Reynolds numbers of 100

Low
Pass
Filter
 $F_c = 4 \cdot F_n$

PD Controller
(acts on
PCD model)

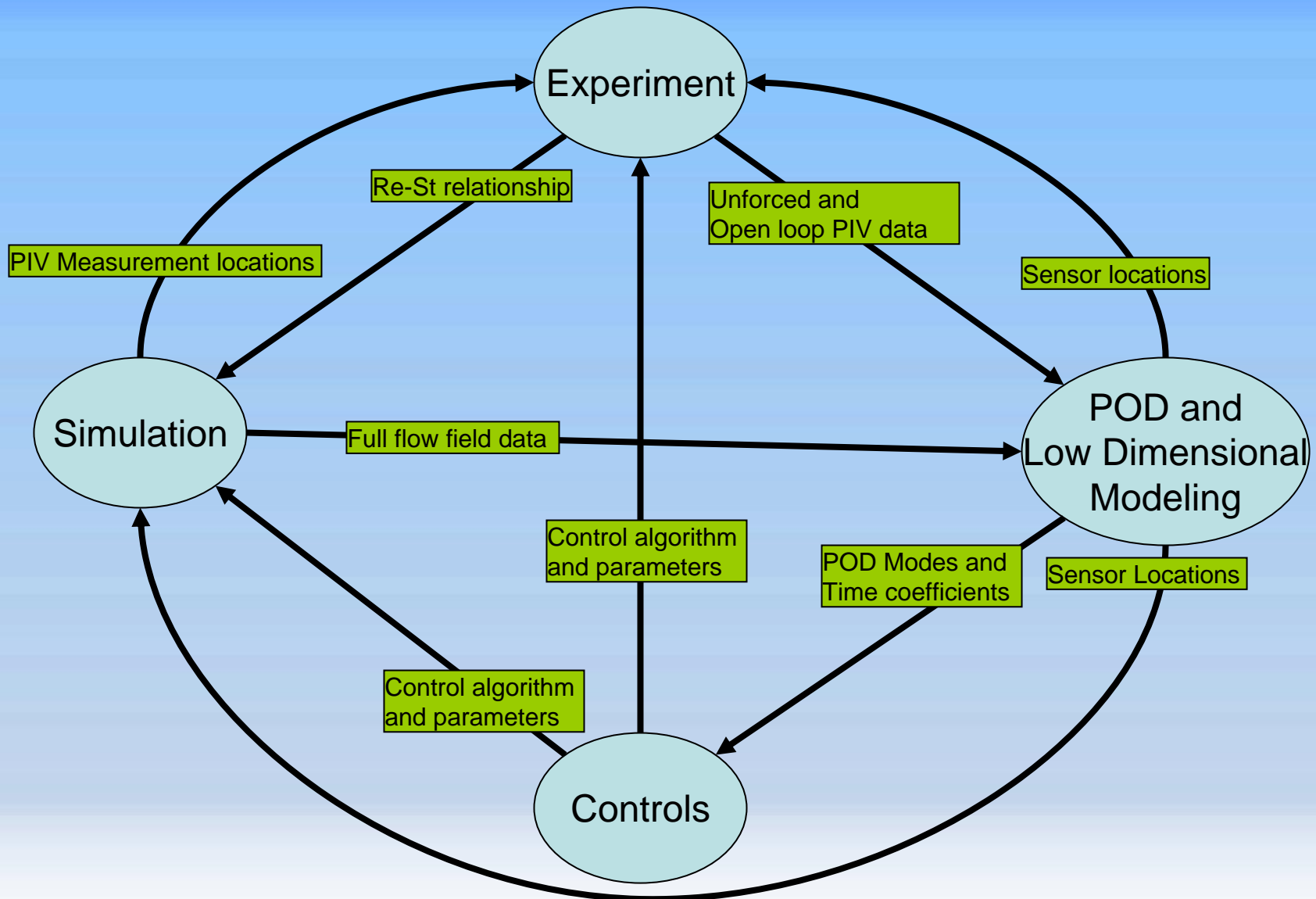
Model
Estimator
(Least
Square)



Sketches from: Munson, Young, Okiishi. *Fundamentals of Fluid Mechanics*. p 601.

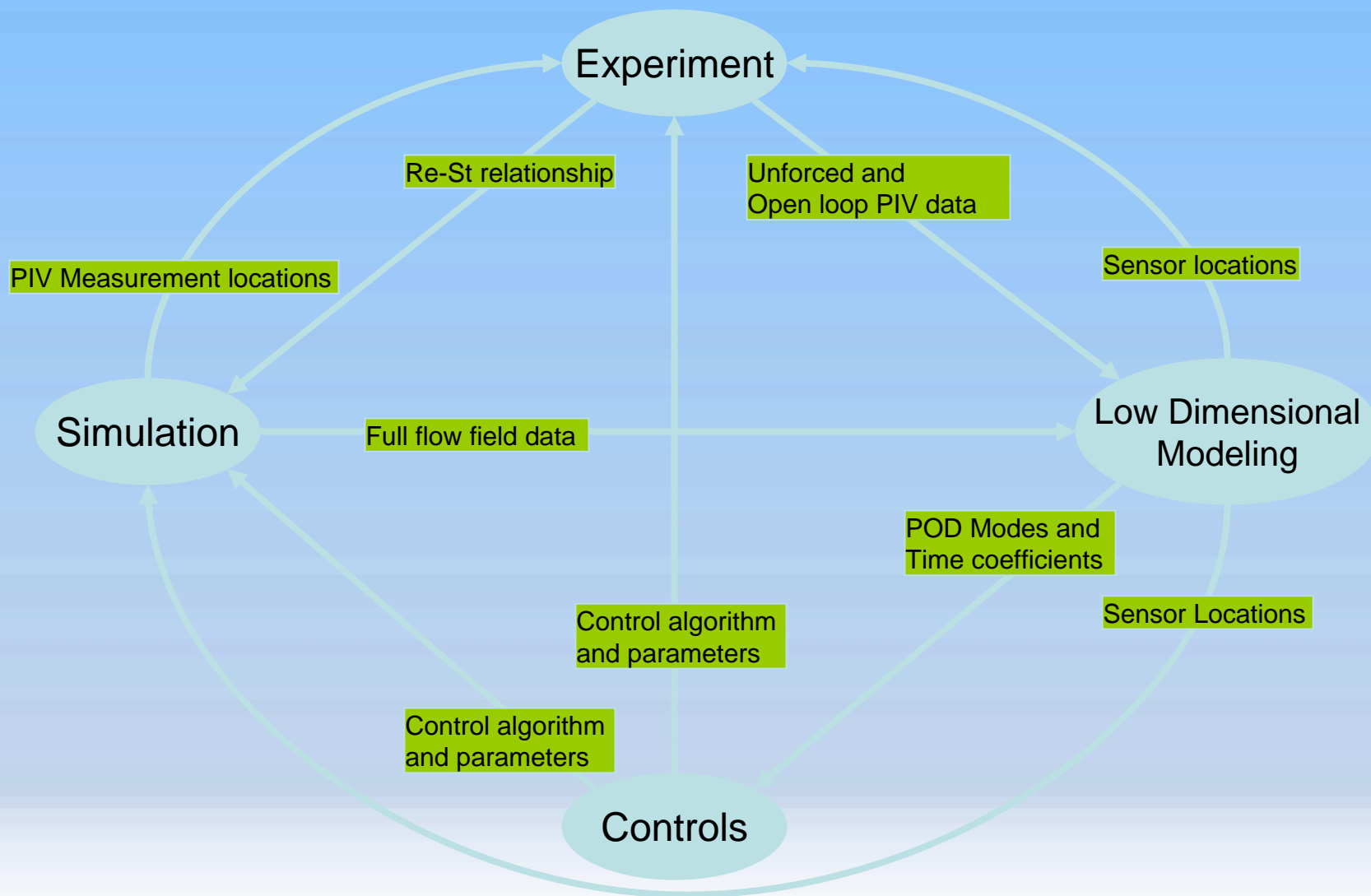


Collaborative Research



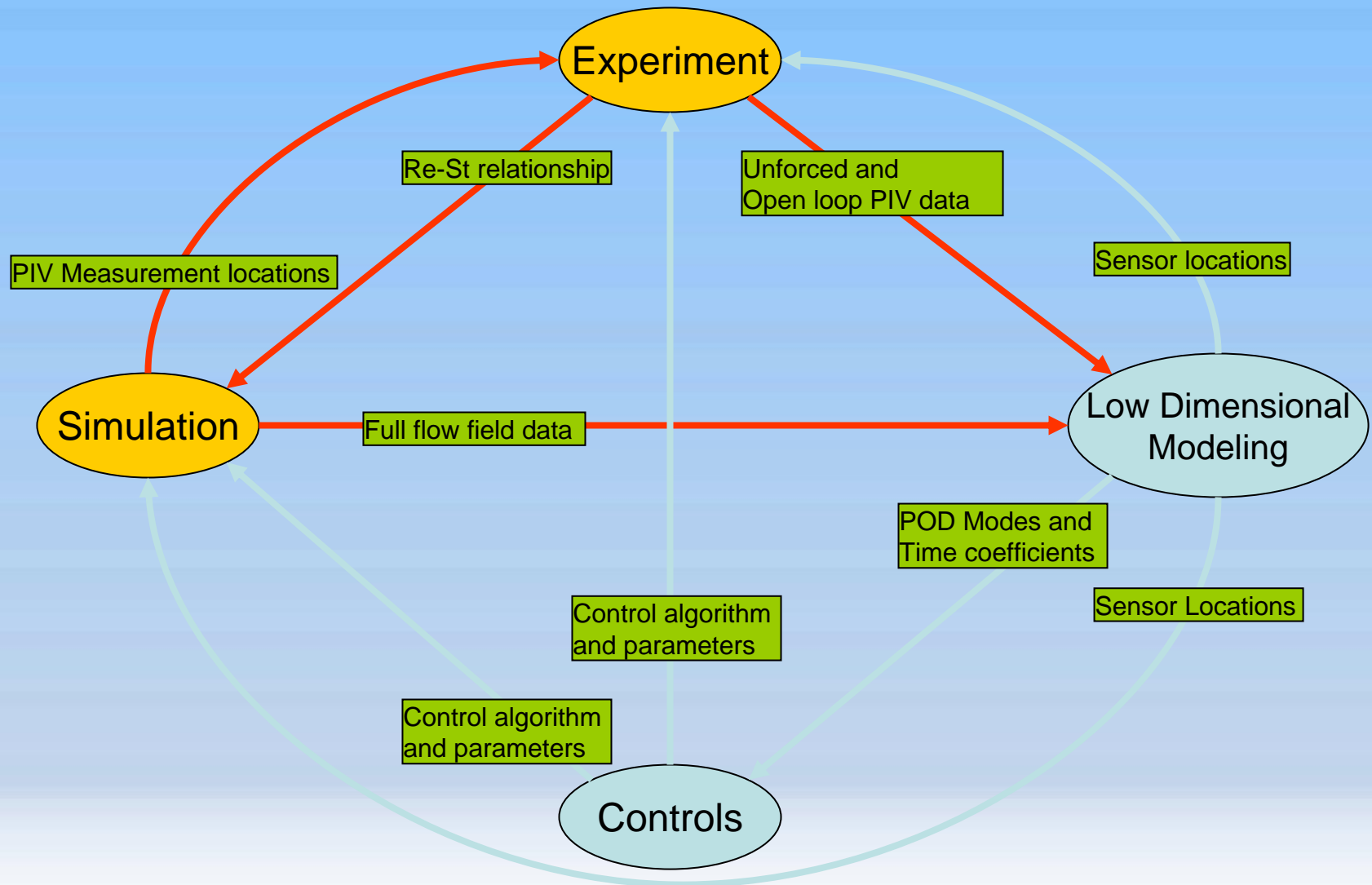


Collaborative Research



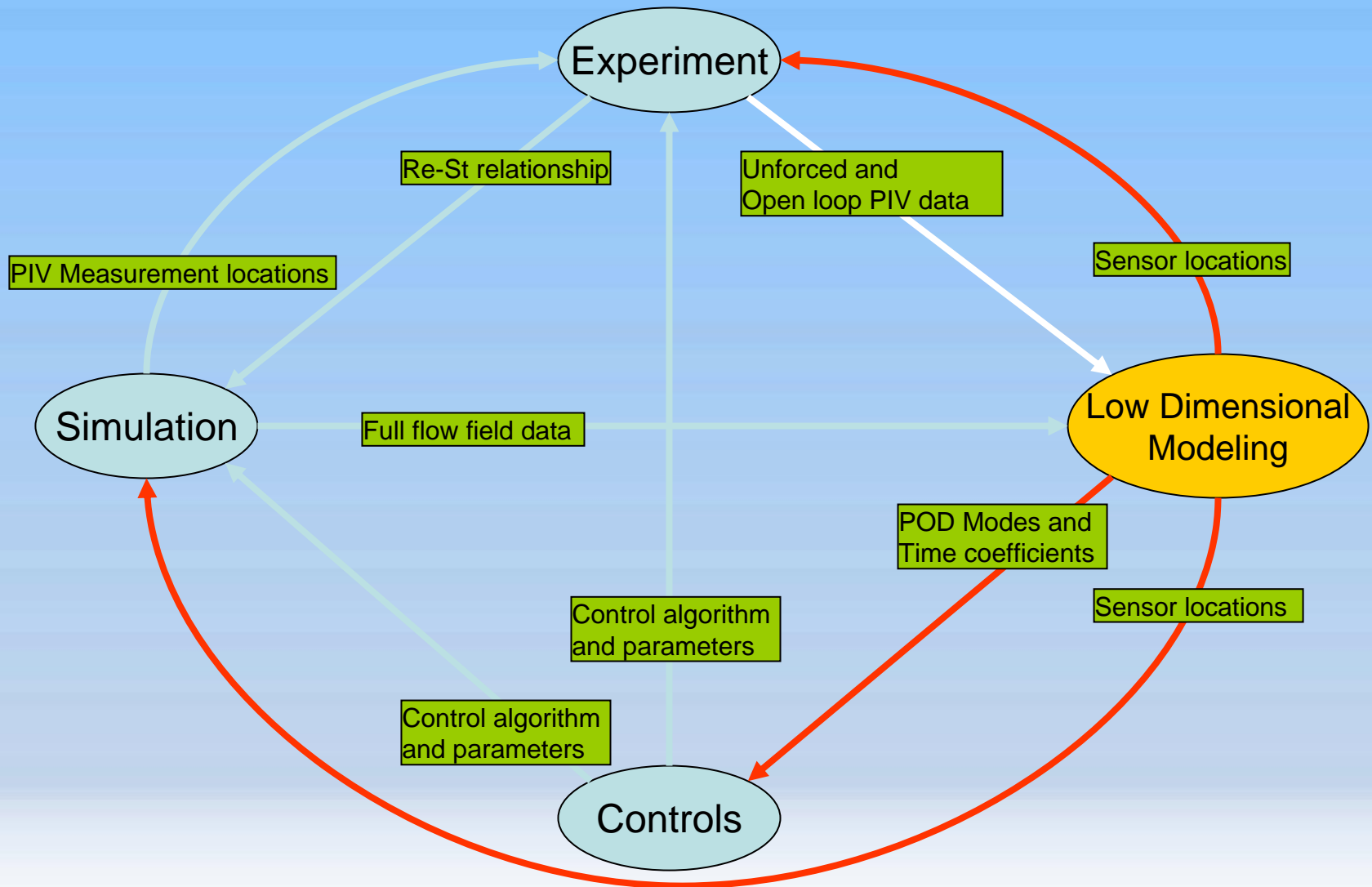


Collaborative Research



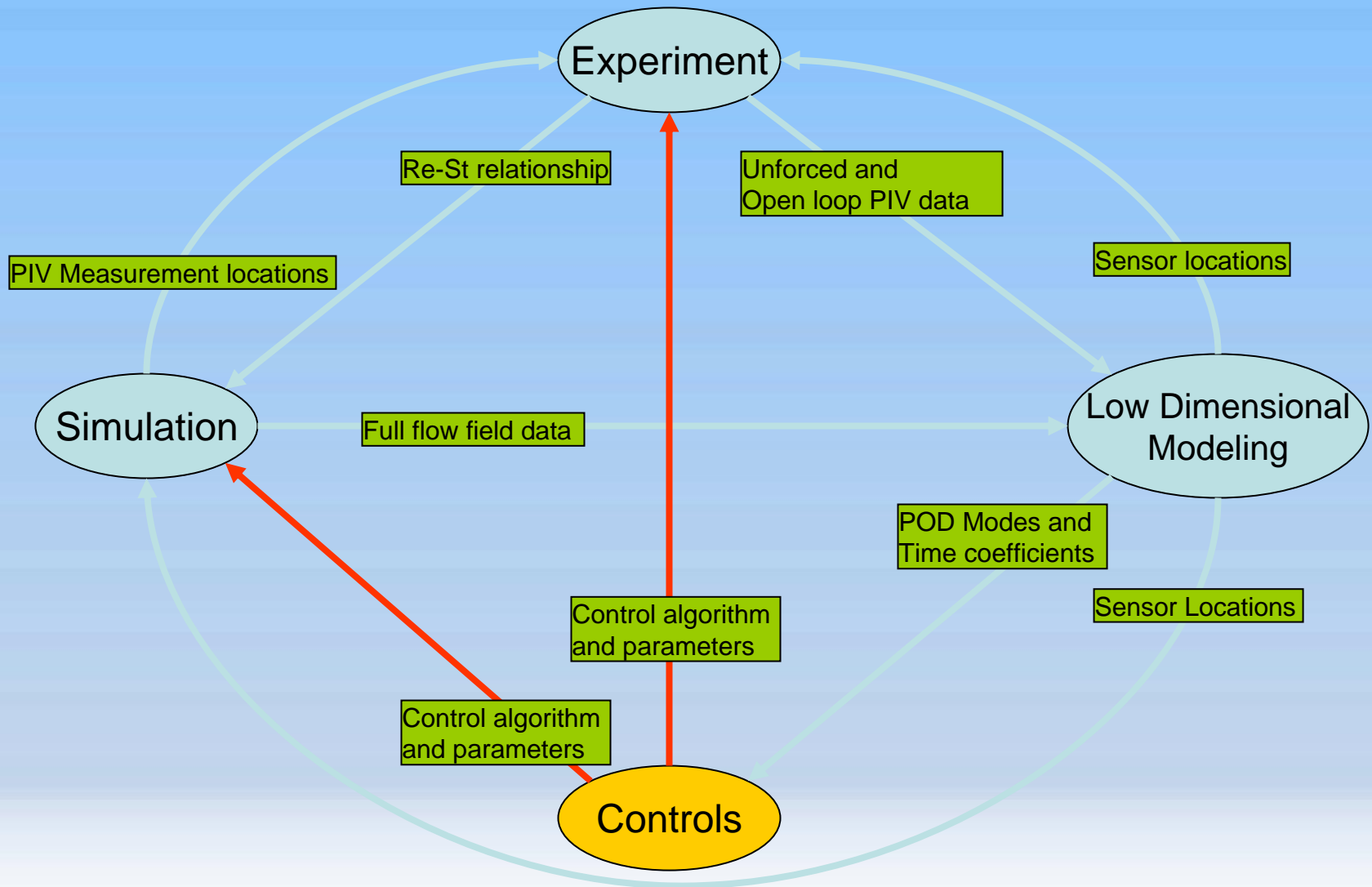


Collaborative Research





Collaborative Research

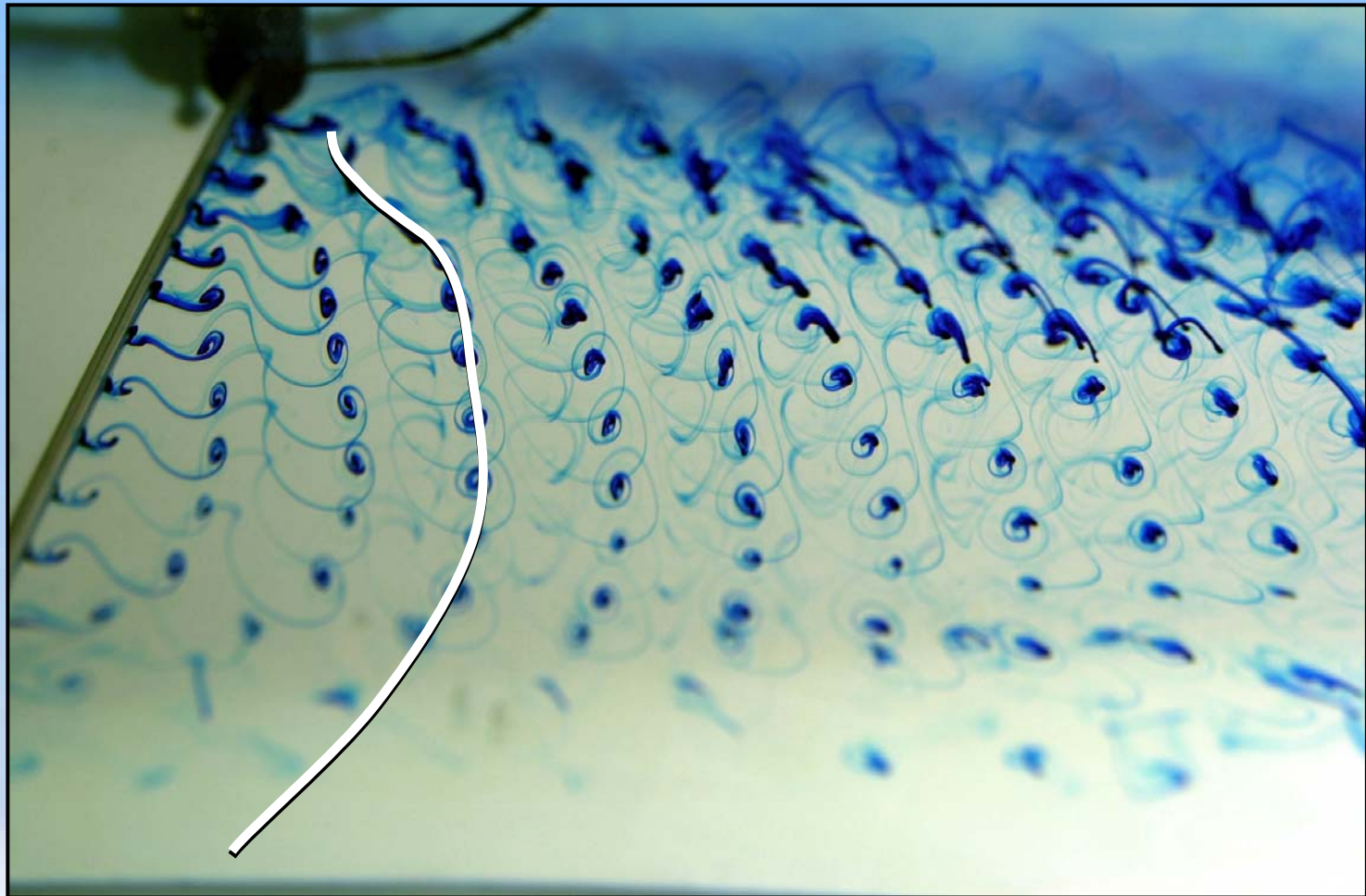




Unforced and Open-loop forced **EXPERIMENTS**

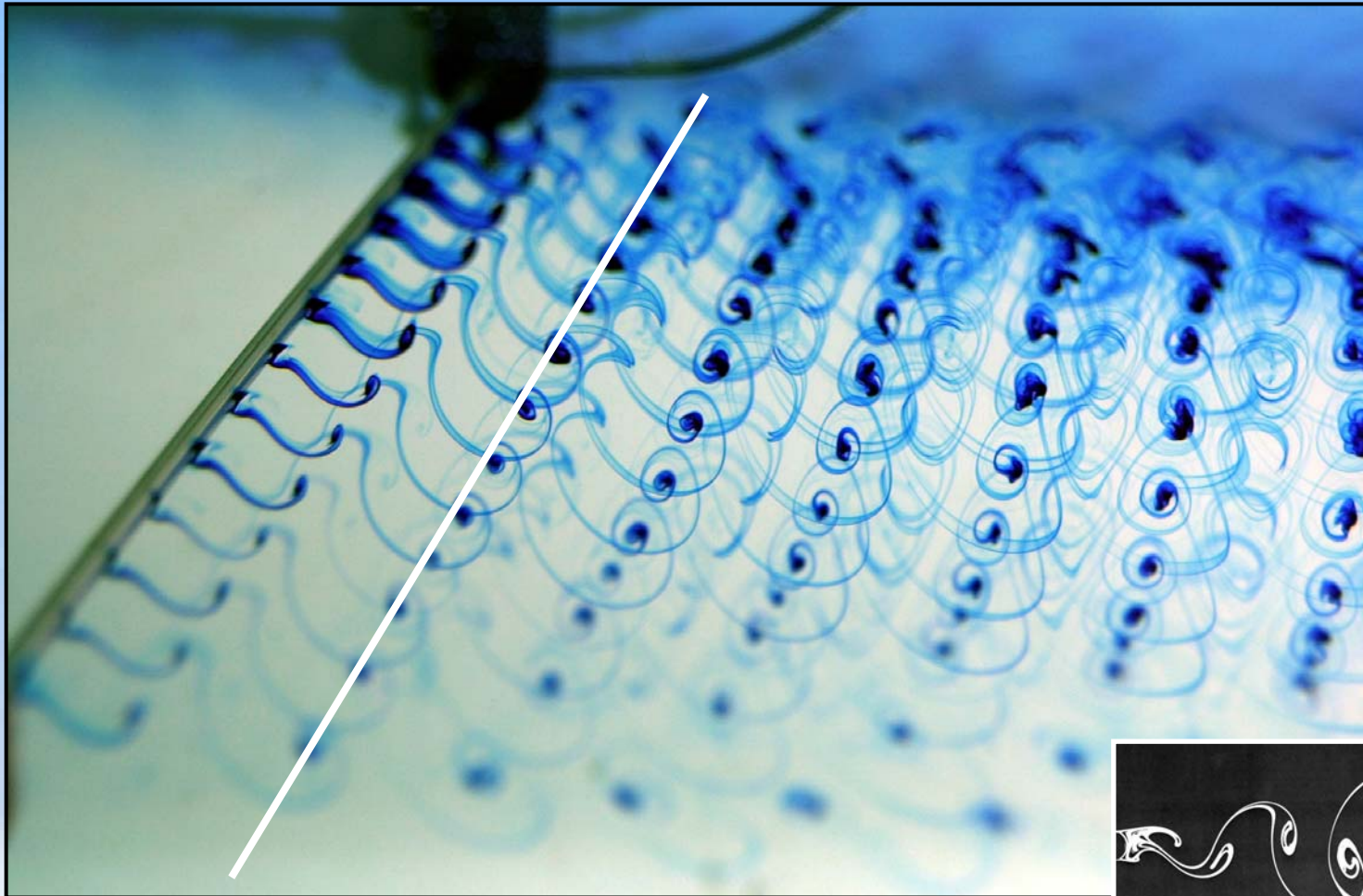


Unforced





Forced case 1

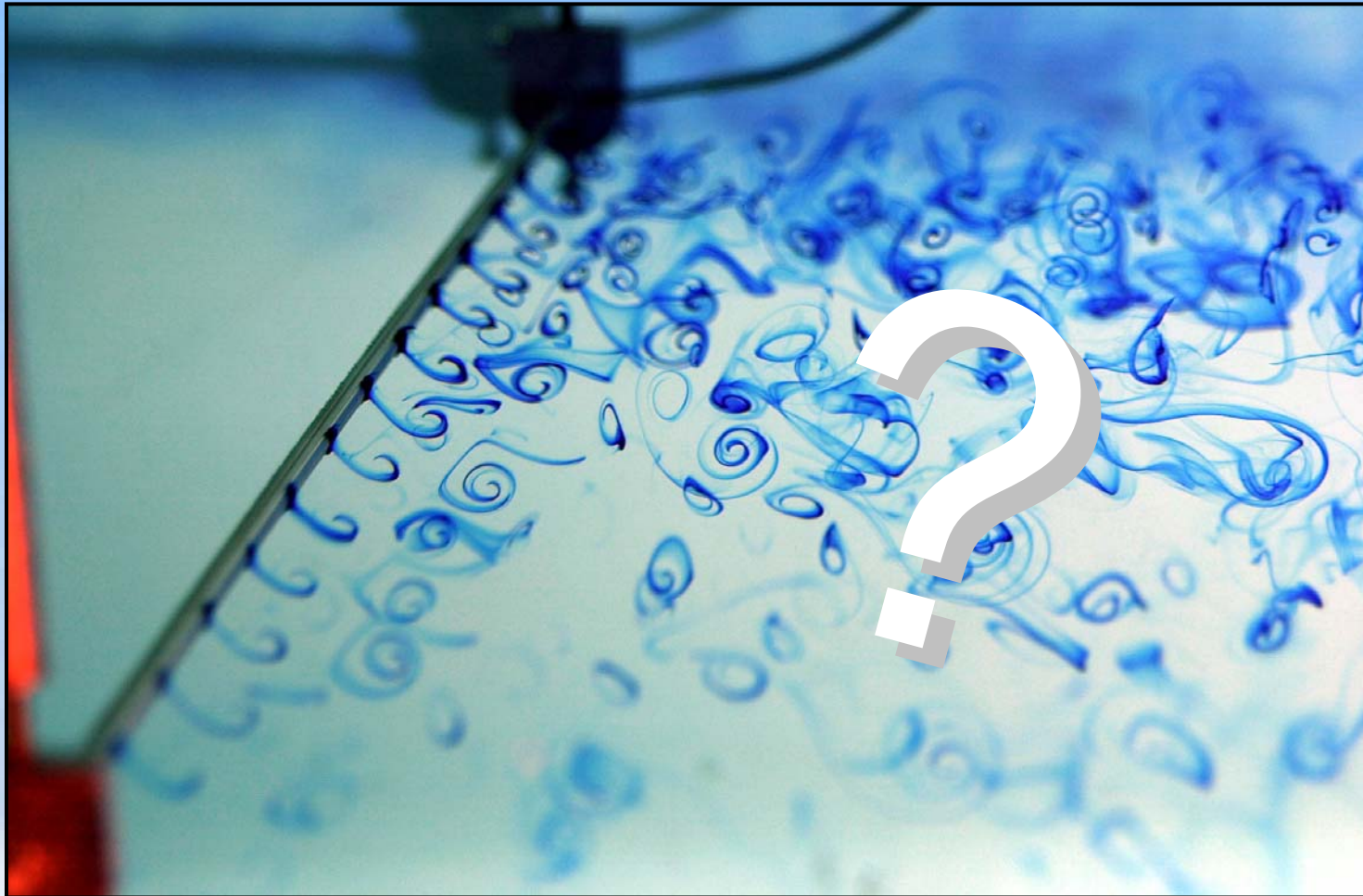


$St/St_n=1$, $A/D=20\%$

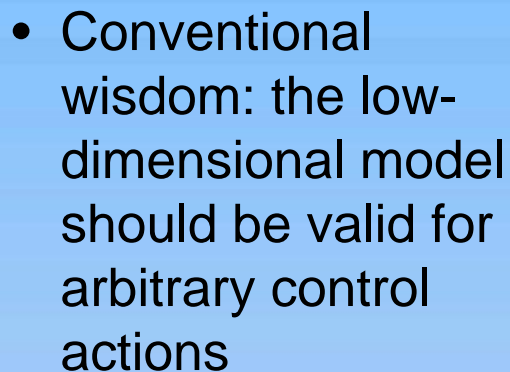




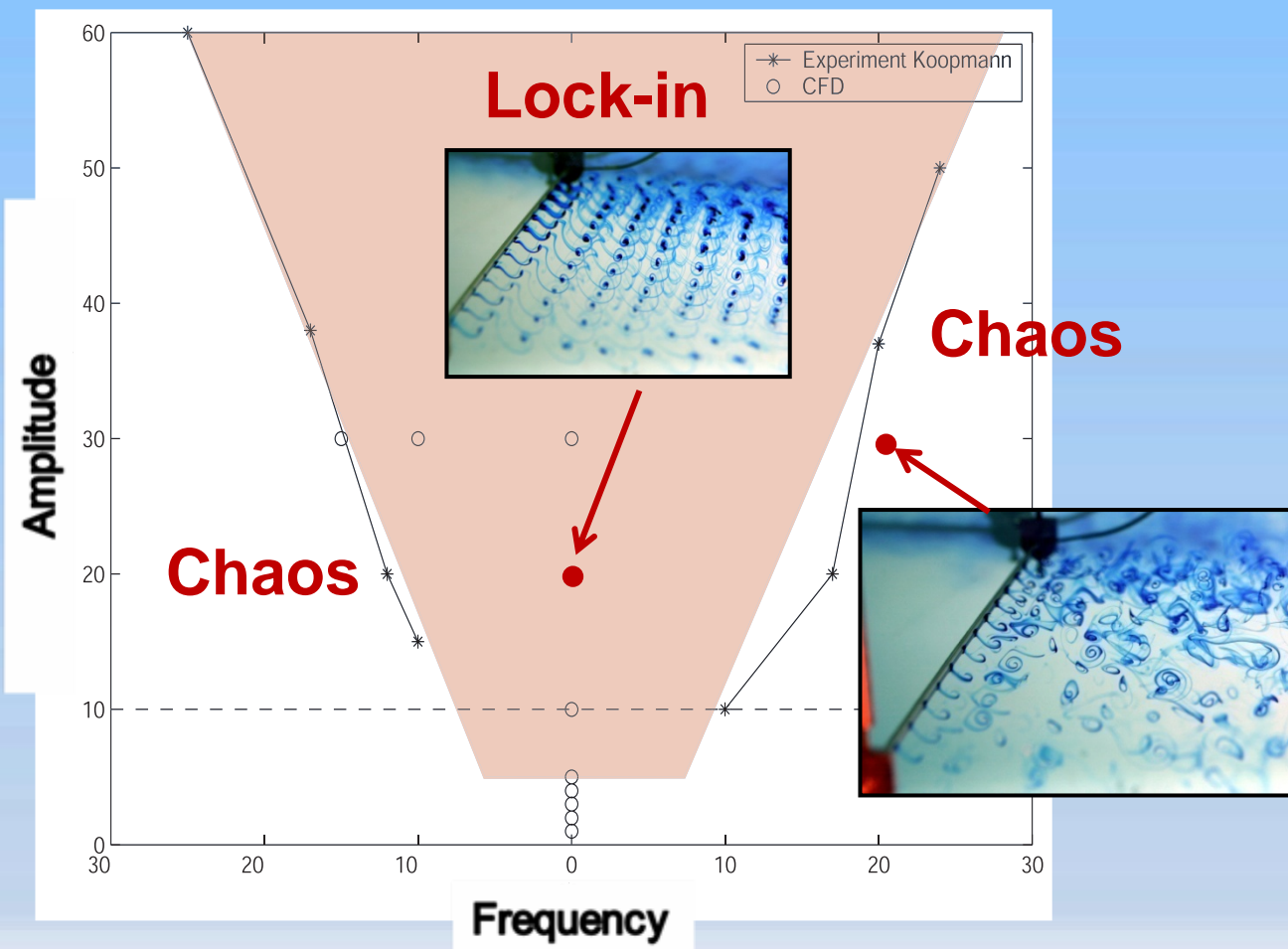
Forced case 2



$St/St_n=1.26$, $A/D=30\%$



- There is a limited envelope of amplitude/frequency of the disturbance wherein the control is effective.





Experiments: Pros and Cons

- Pros

- Overview/visualization of flow field
- Easy scan of frequency/amplitude parameter space
- Final verification

- Cons

- Expensive model design and building
- Limited data available
 - Velocity, pressure
 - Field of view
 - State-of-the-art (e.g. PIV) only 2D



Unforced and open-loop forced

SIMULATIONS



Simulations

- Cobalt

- Hybrid-Unstructured, Compressible Solver
- Point Implicit with Subiteration
- 2nd-Order Temporal and Spatial Accuracy
- Turbulence Models
 - RANS: SA, SARC, SST, and others
 - Hybrid RANS/LES: SA-DES, SARC-DES, SST-DES
- Domain decomposition using ParMETIS (Dr. Karypis, UMN)
- MPI parallelization
 - Over 98% efficient on 1024 processors
- Arbitrary Lagrangian Eulerian (ALE) for rigid body motion
- Variety of motion types: 1DOF, 6DOF



- Matlab

- Controller development
- Data analysis
- Post-processing



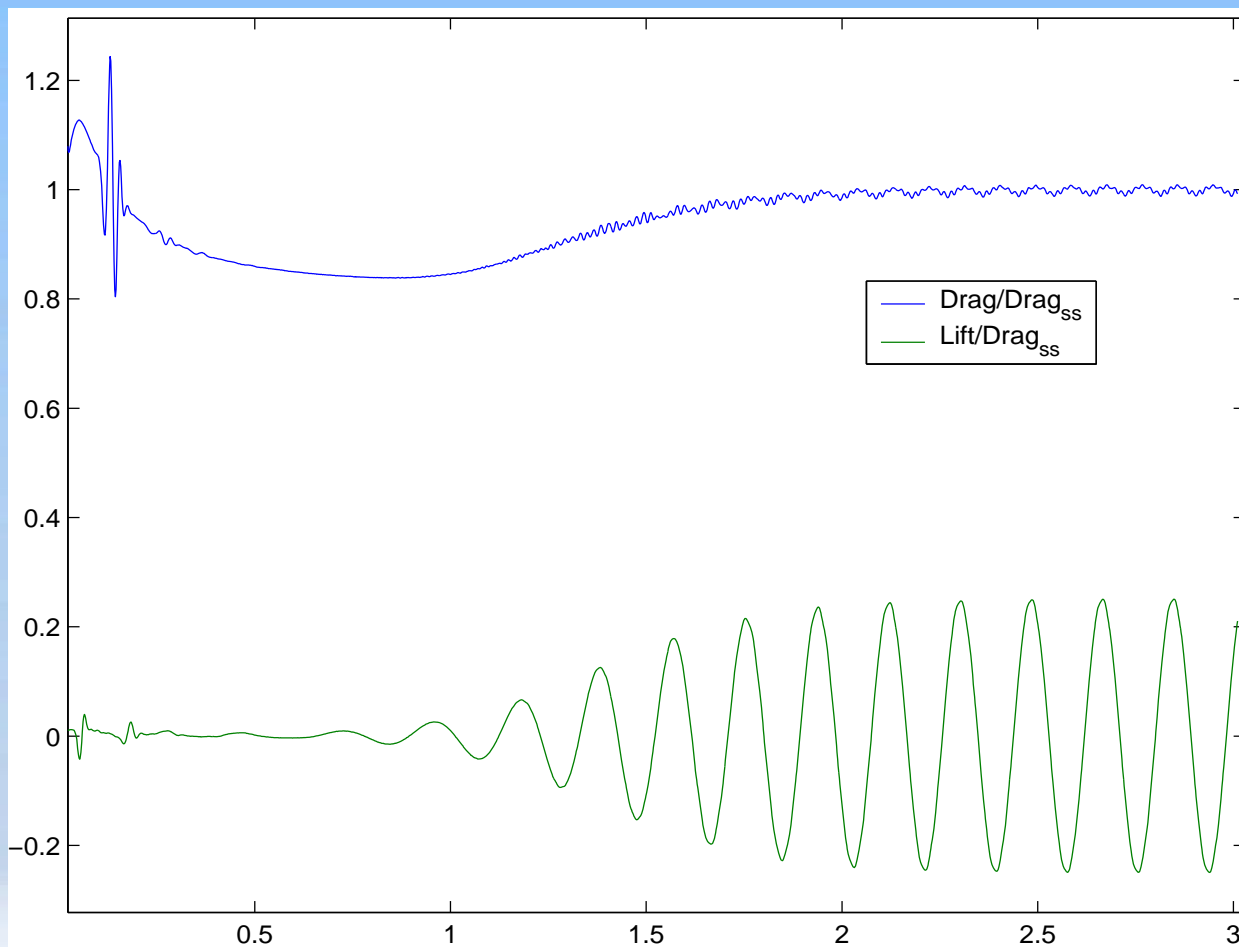
- Cobalt-Matlab interface for feedback flow control

- Developed under current AFOSR STTR Phase I/II
- HDF5 output



Transient Startup Data Set

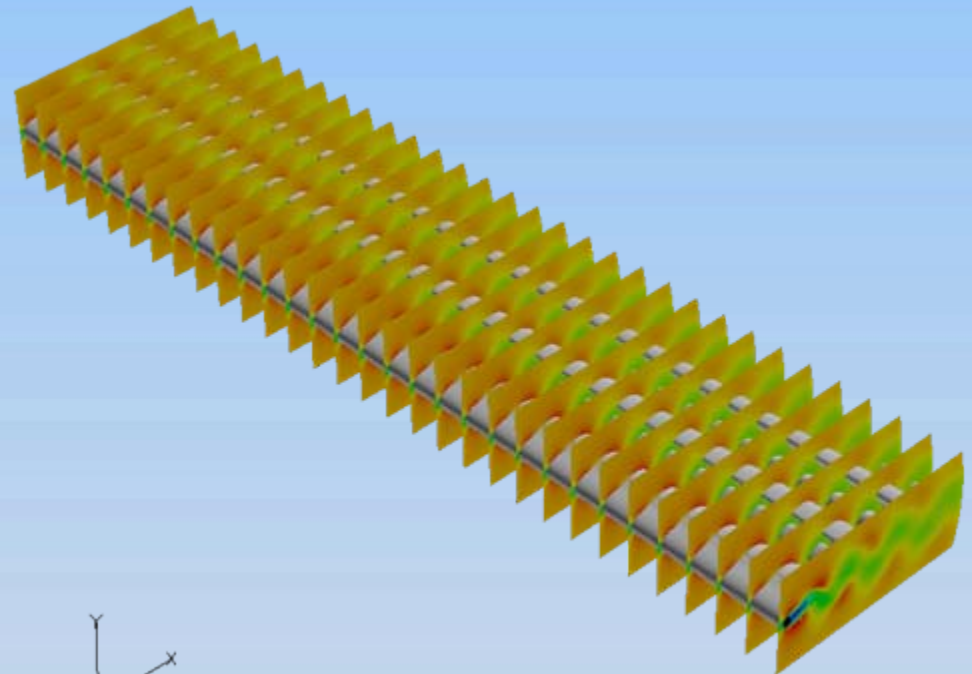
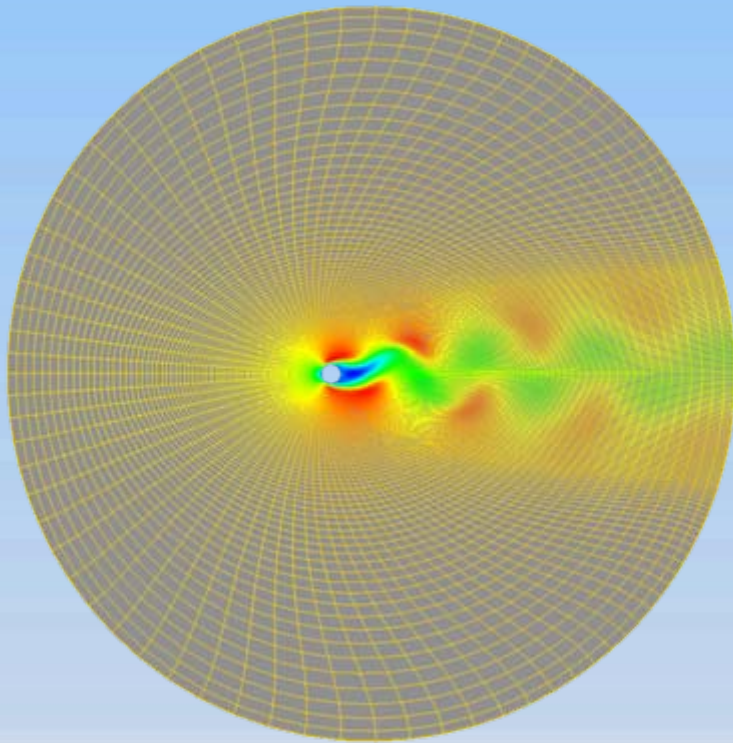
$Re = 300$





3D cylinder: grid

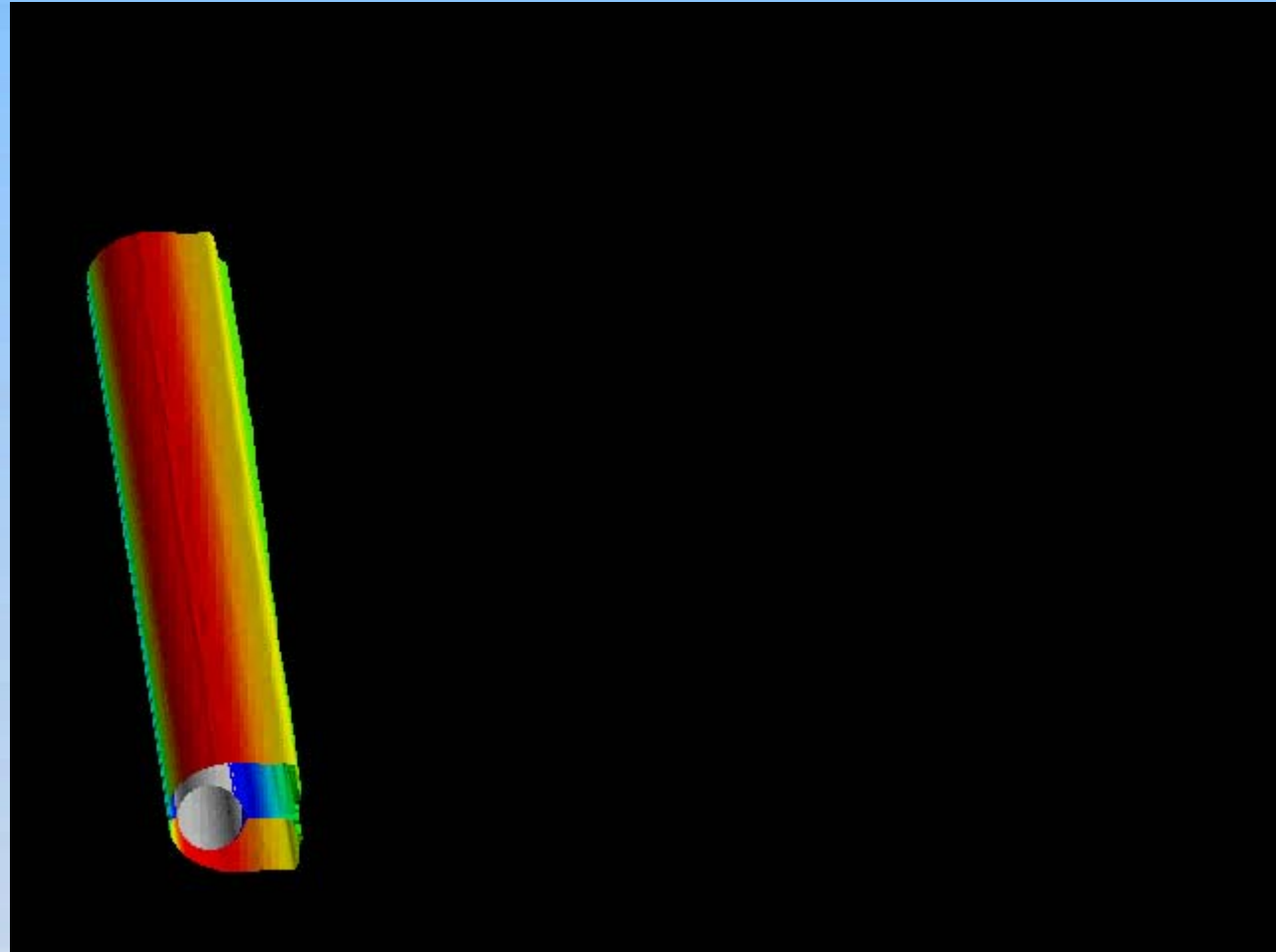
- Body fitted “O” grid extruded along cylinder axis
- 163x198x31 (r, q, z) points





Simulation, $Re=100$

- $L/D=96$
- Grid:
 - 2M nodes
 - 31 spanwise planes
- Time:
 - 50 periods
 - 5.6CPUh/period

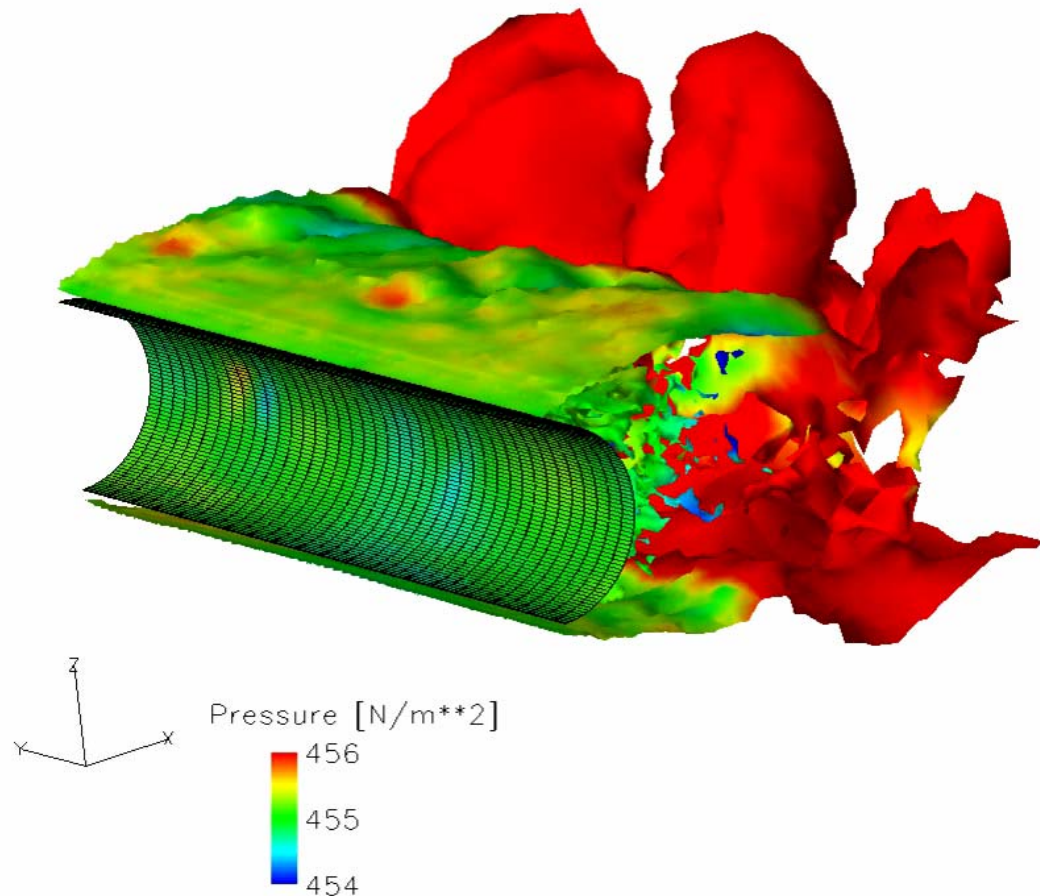




Simulations, $Re=20,000$

- $L/D=4$
- Grid:
 - 1M nodes
 - unstructured
- Time:
 - 180CPUh/period

Isosurface of total vorticity, $W=20 \text{ 1/s}$
Time=1.152s





Simulations: Pros and Cons

- Pros

- Detailed flow field information
 - Time
 - Space
- Range of possible flow conditions
 - Reynolds number transients

- Cons

- Time consuming model/grid design and building
- Time consuming data generation
- Limited number of conditions possible
 - Parameter space



MODELING AND CONTROLS



Control of a Ginzburg-Landau cylinder wake model



- The complex Ginzburg-Landau (GL) equation model

- vortex dynamics in bluff-body (such as a circular cylinder) wakes

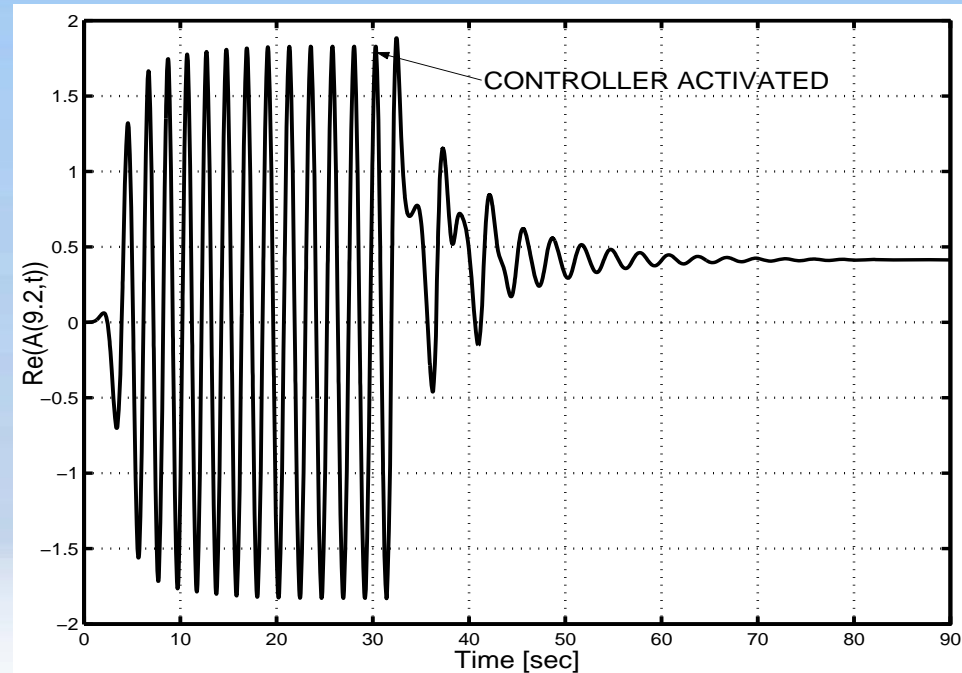
$$\frac{\partial A}{\partial t} + U \frac{\partial A}{\partial x} = \mu(x) A + (1 + jc_d) \frac{\partial^2 A}{\partial x^2} - (1 + jc_n) |A|^2 A + F(x, t)$$

- Wake stability of the GL model is defined by the growth parameter $\mu(x) = \mu_o + \mu'x$

- μ_o is similar to a Reynolds based on the cylinder diameter.
 - For $\mu' < 0$, the stability features similar to 2D cylinder wake.

Condition Studied	C_{11}	C_{12}	C_{21}	C_{22}
12.5% Above Critical	1.4524	4.1250	5.1486	4.1643
20.0% Above Critical	4.2744	6.9612	4.7633	0.1911

Look-up table for Coefficients C_{ij} of the modal estimator



Wake Signal at 20% above Critical



Proper Orthogonal Decomposition

Flow field from experiment or simulation



N Snapshots of
Flow Field
 $U(x,y,t)$
 $V(x,y,t)$

POD
(Proper
Orthogonal
Decomposition)

K Temporal Mode
Amplitudes

$A_1(t)$

$A_2(t)$

...

$A_K(t)$

K Spatial Modes

$U_1(x,y)$

$V_1(x,y)$

$U_2(x,y)$

....

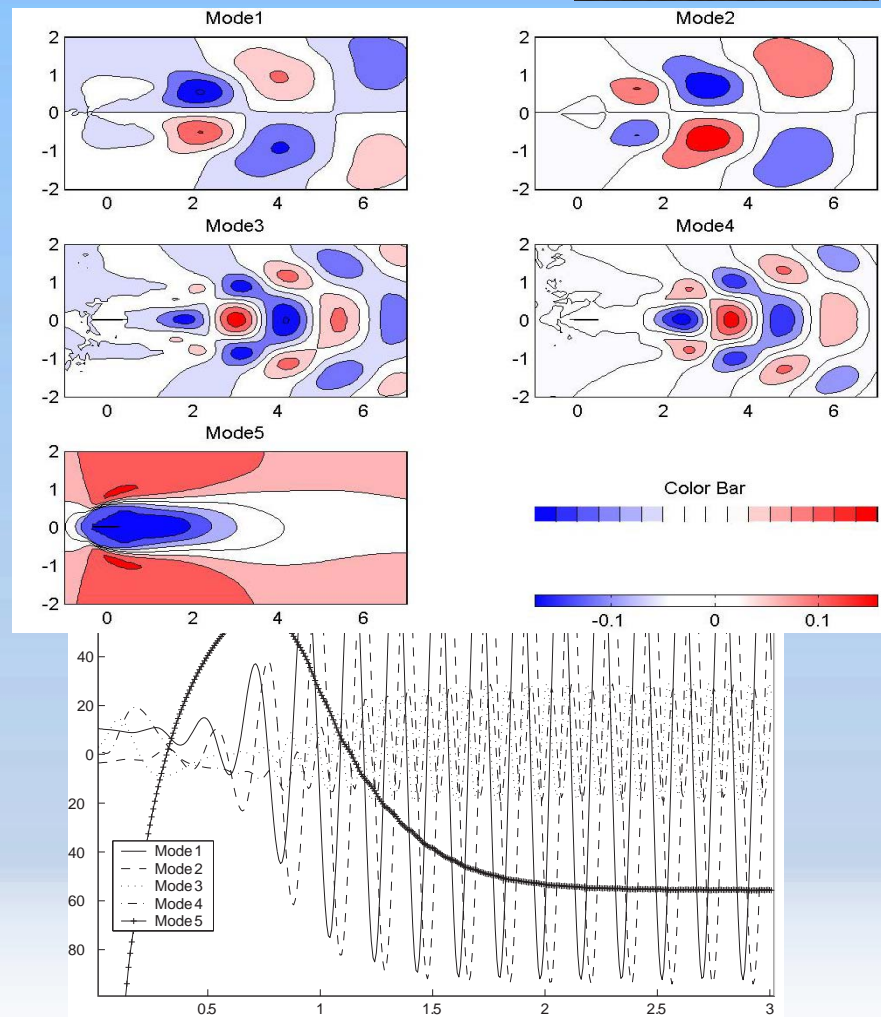
$V_K(x,y)$



POD and Low Dimensional Modeling

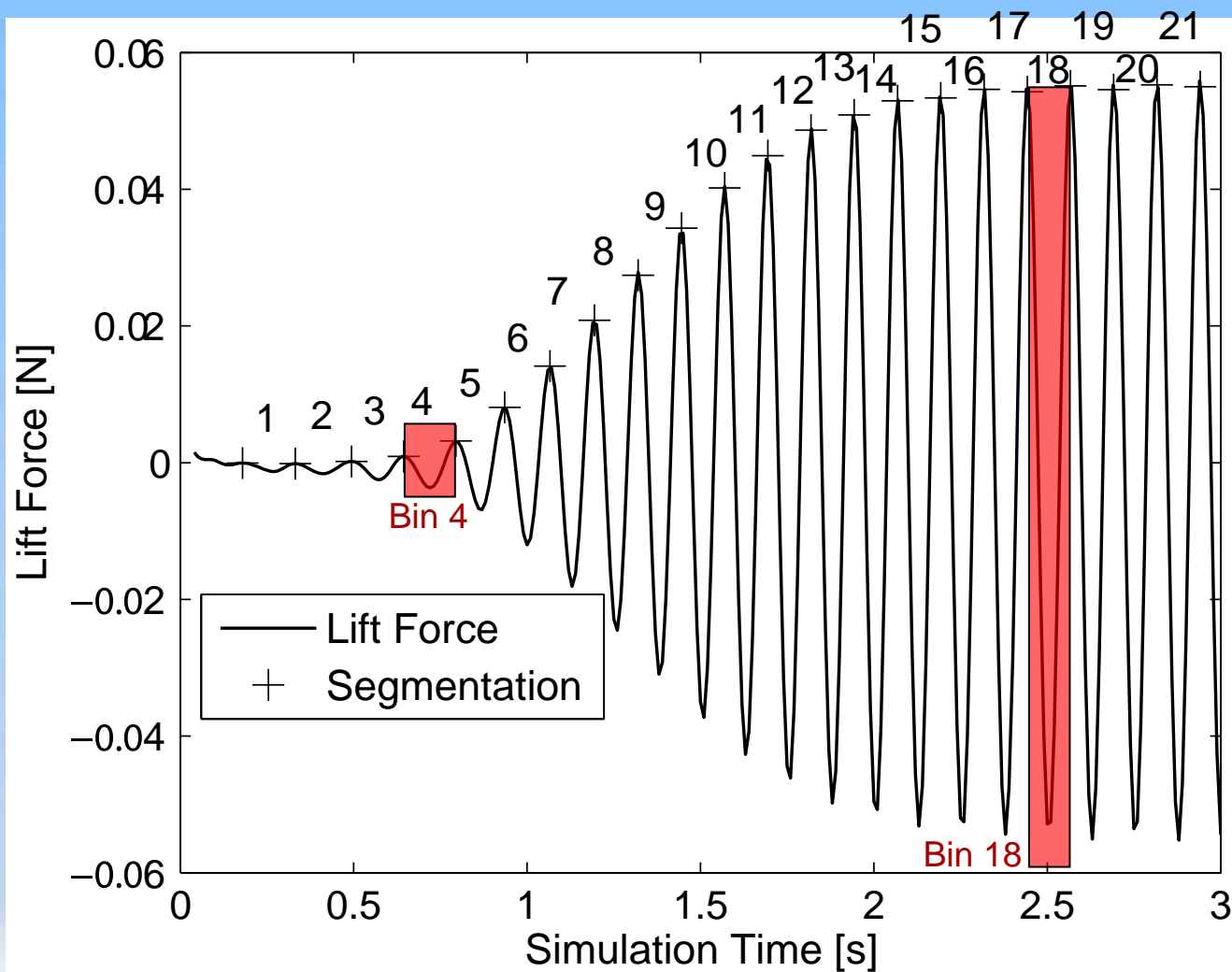


- POD spatial modes
 - Flow characteristics
 - Sensor placement studies
- POD mode amplitudes
 - Low Dimensional Model Development
 - Develop real time nonlinear mapping based on neural networks between measurable quantities (pressure, velocity) and low dimensional states
 - Linear and nonlinear system identification tools
 - Develop control strategies
- Reconstruction of the flow field possible
- Massive reduction of CFD simulation data



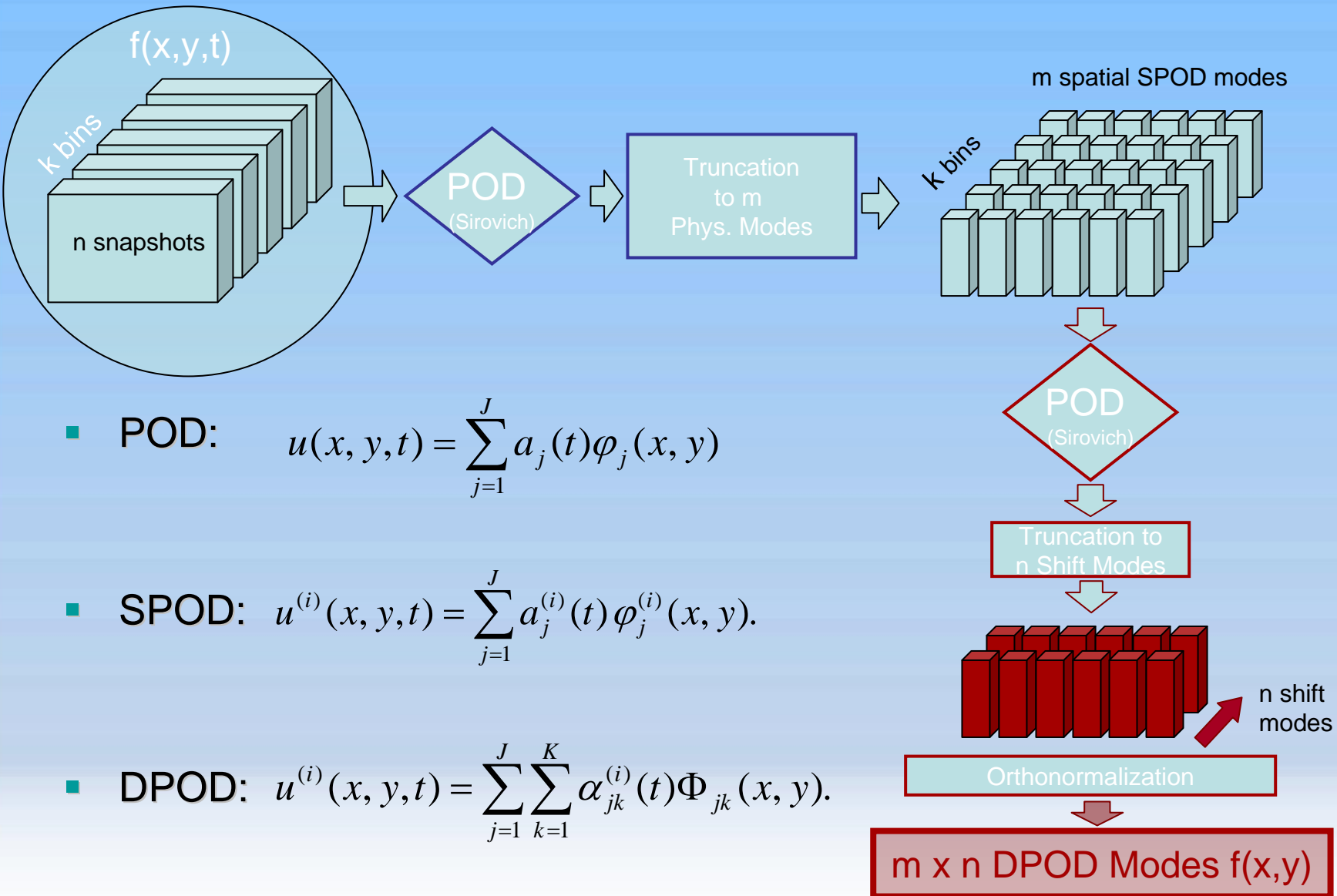


SPOD Segmentation





DPOD Basis Construction





Modeling: Pros and Cons

- Pros

- Rapid exploration of controller parameter space
- Implementable in real-time with relative ease
- Effectively targets the large coherent structures in the flow

- Cons

- Model building is tough
- Quality/validity of model depends on underlying data
 - Parameter space



Feedback controlled

EXPERIMENTS



Experimental setup

Flow Vis Camera

Cylinder Actuation System

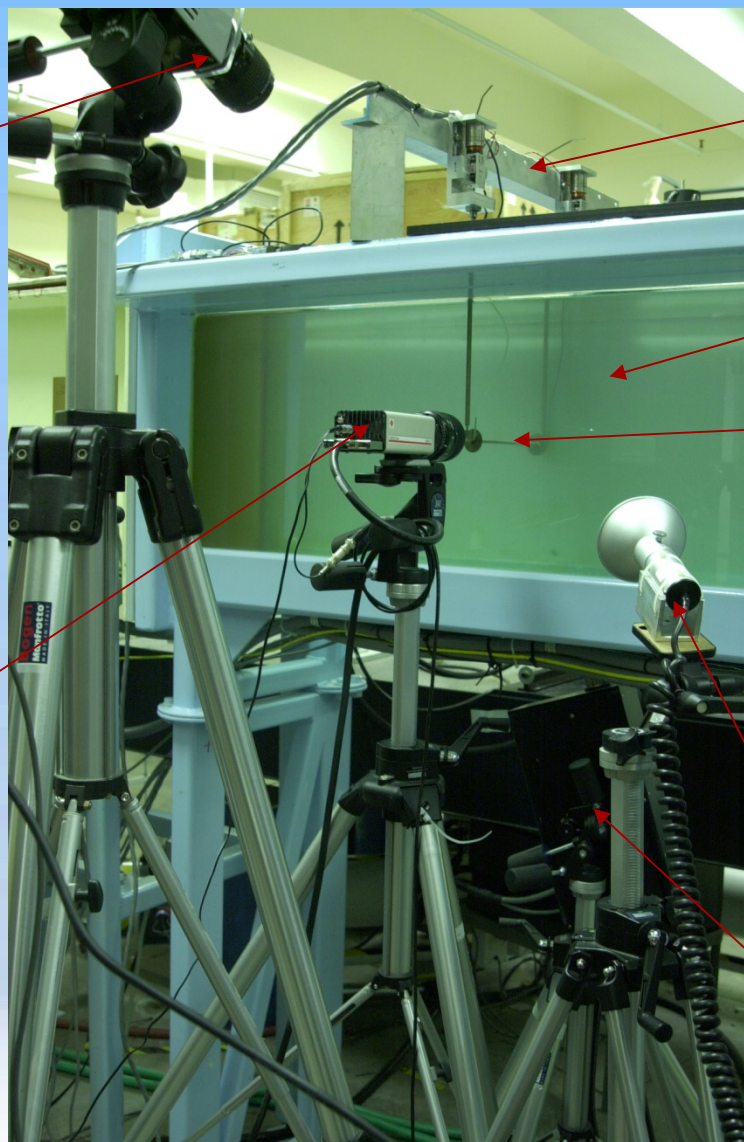
Test Section

Cylinder Model

PIV Camera

Strobe Light

PIV Laser





Sensor and Flow Vis Setup

Dye Ports for Flow Vis
(@ far end of model)

Laser Light Sheet and Feedback
Measurement
Plane
(@center of model)

Flow Direction

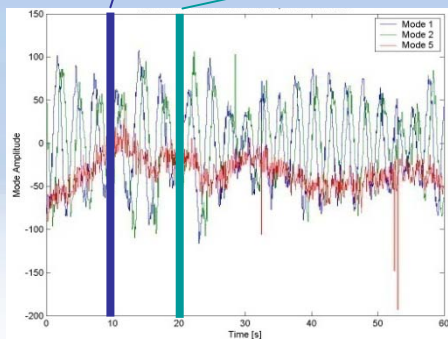
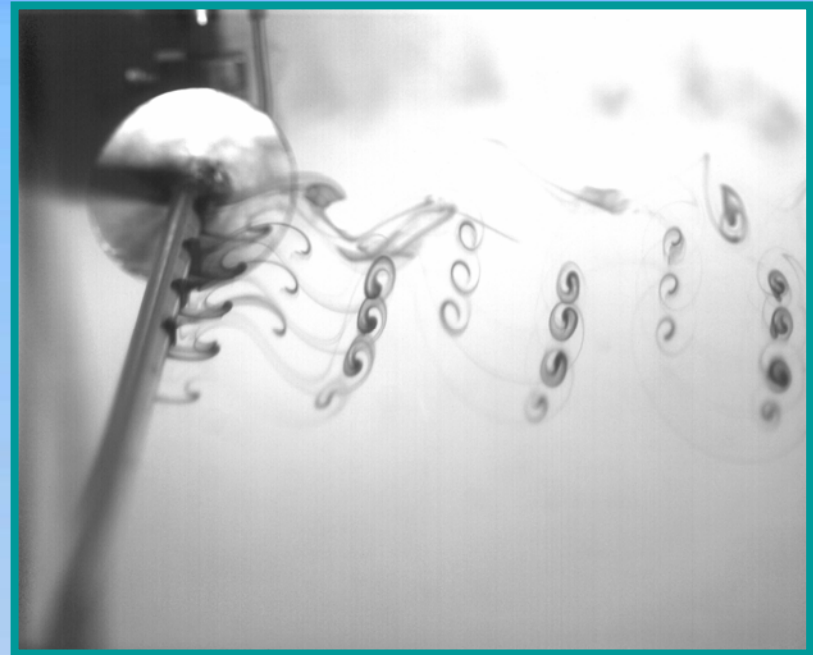
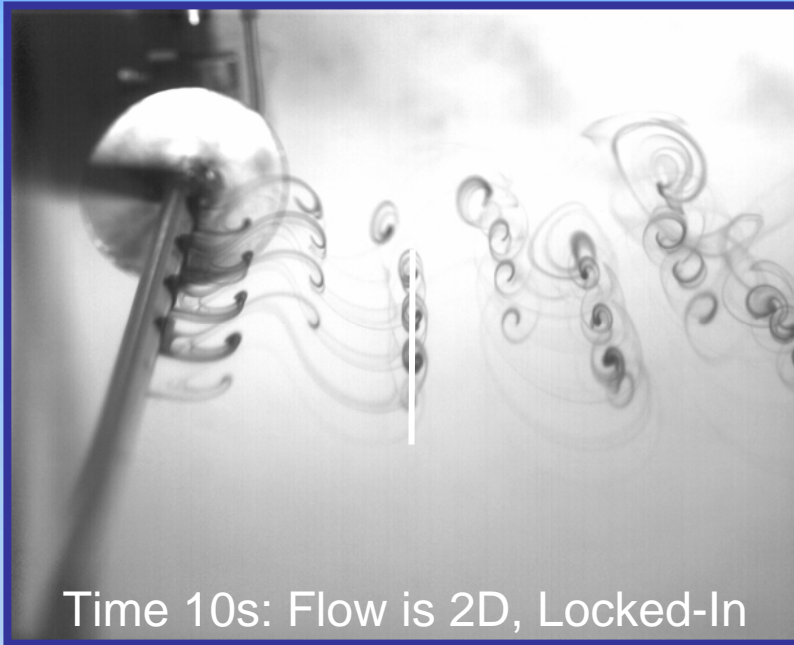
Cylinder Model

PIV Laser



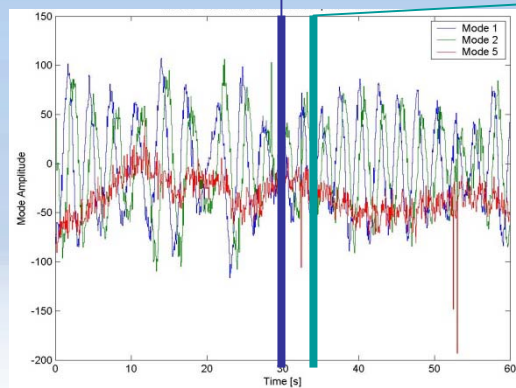
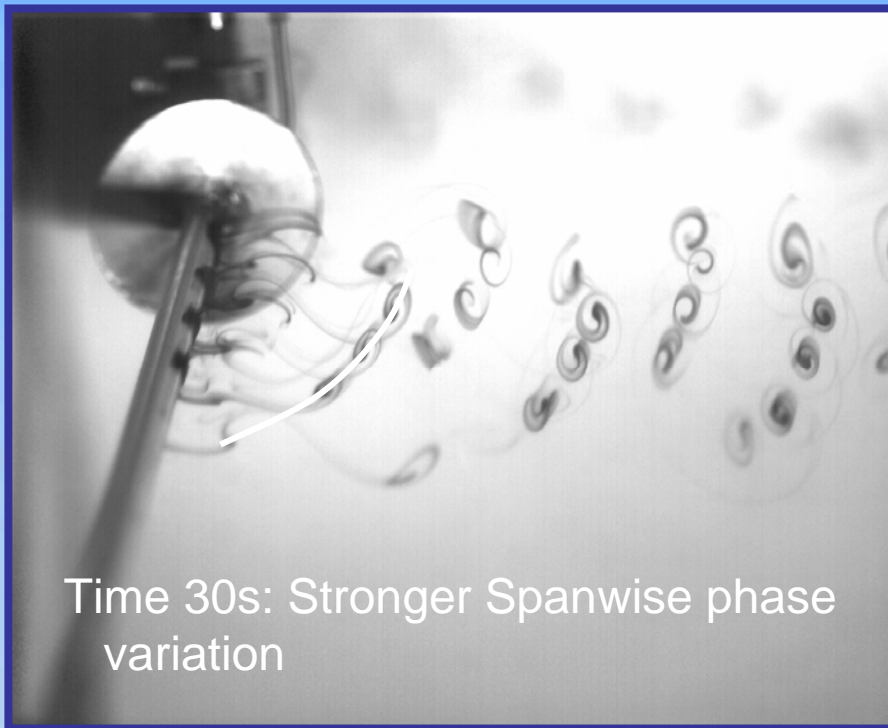


Phase 115° Flow Vis





Phase 115° Flow Vis – Cont'd





Experiments: Pros and Cons

- Pros

- Verification of entire feedback flow control concept
- Visualization of flow field
- Easy scan of controller parameter space

- Cons

- Expensive feedback control implementation
- Limited data available
 - Sensors for real time feedback are limited to 2D
 - Wake
 - Surface
 - 3D information only available through flow visualization (offline)
- Debugging of feedback controller difficult

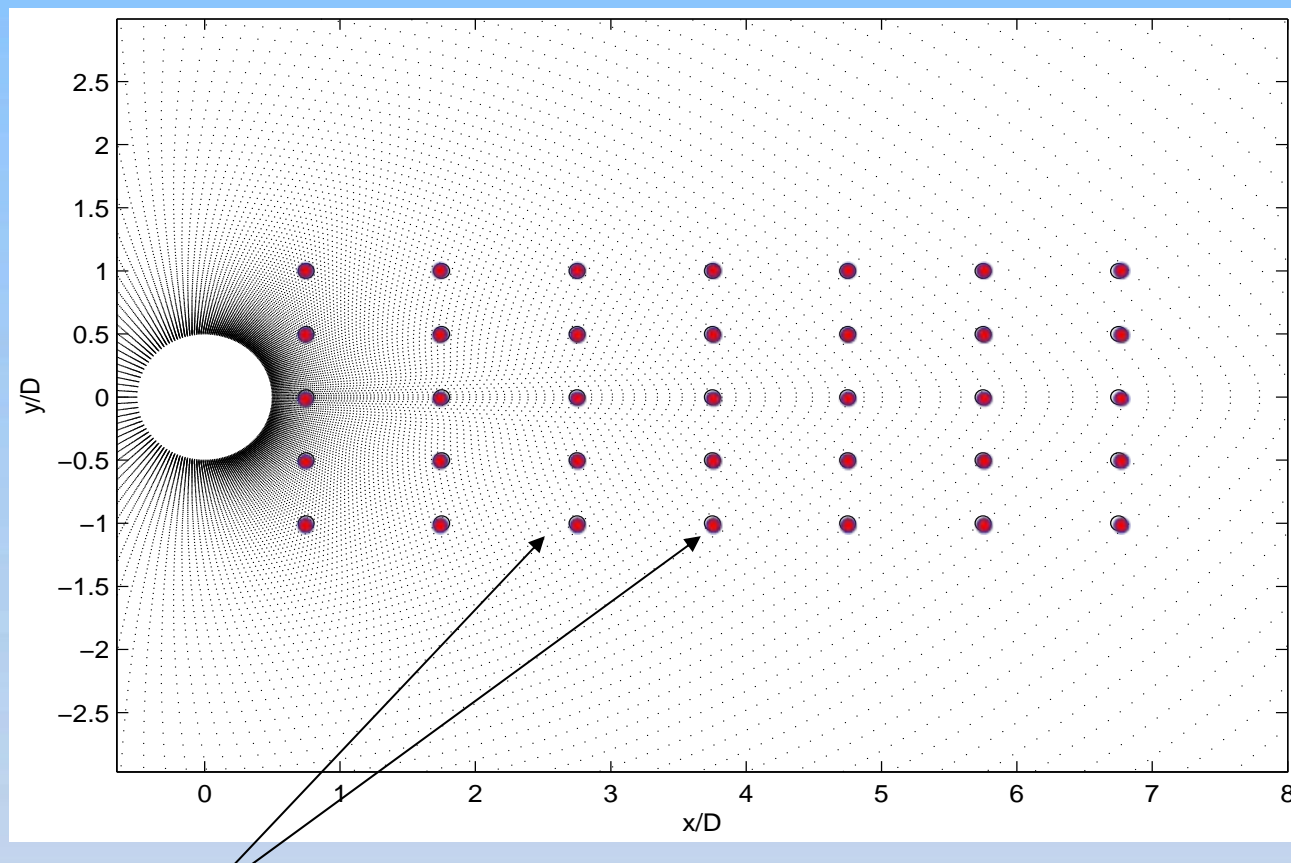


Feedback flow control

SIMULATIONS



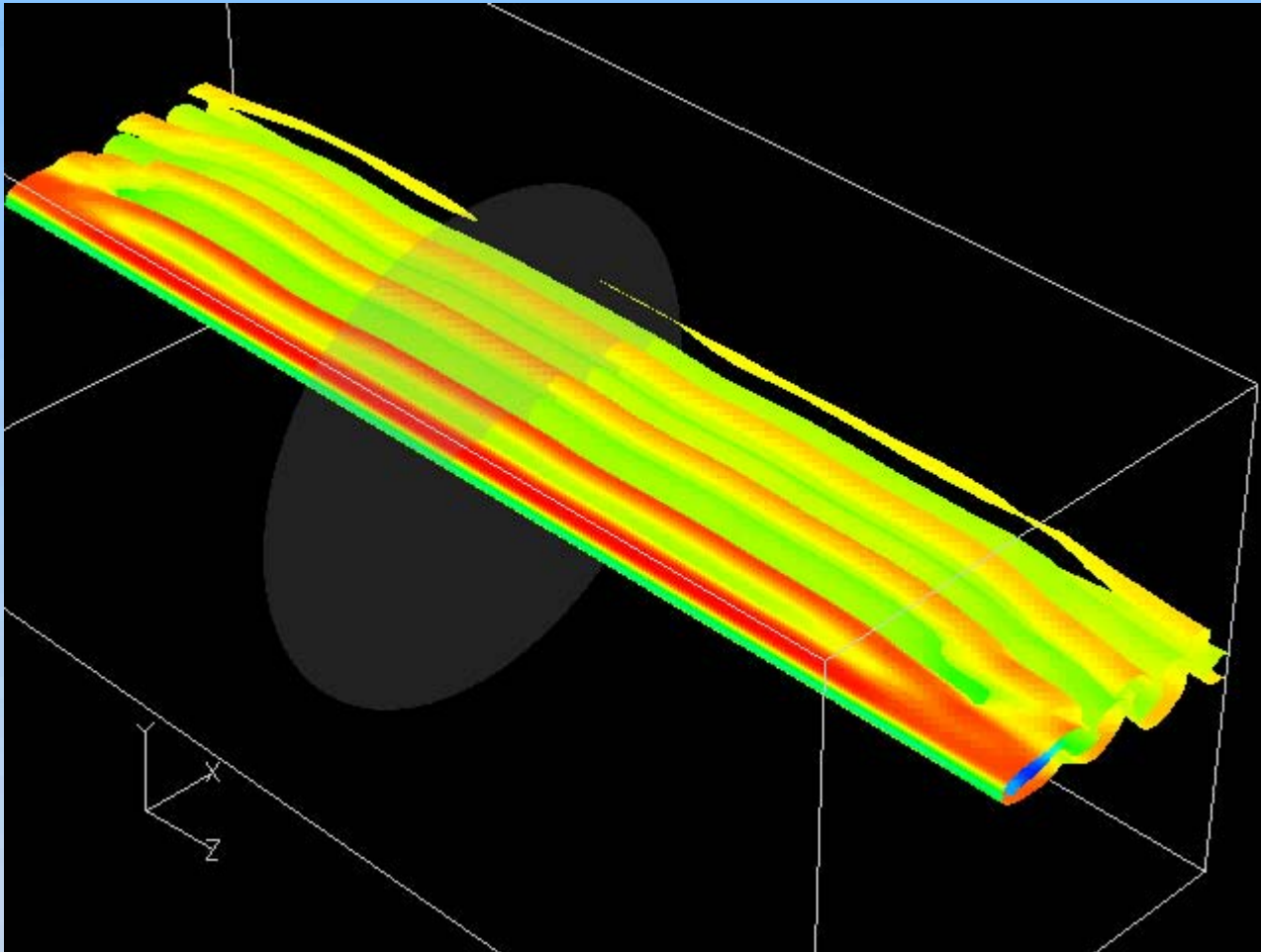
3D cylinder: wake sensors



35 Sensor Locations



Feedback controlled

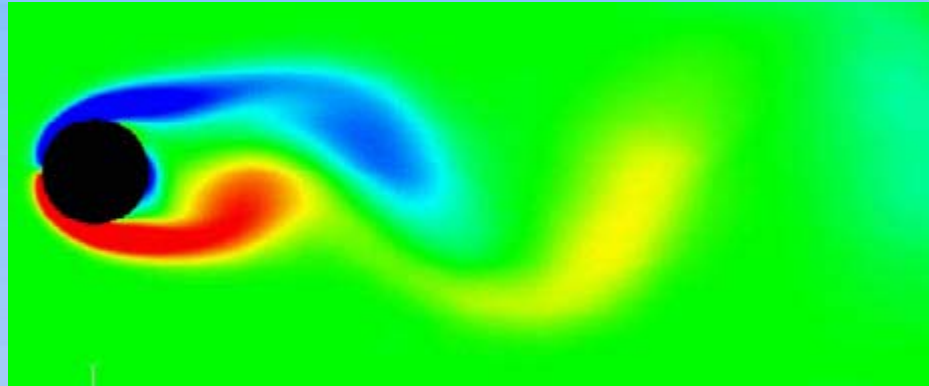


Isocontours of Vorticity colored by U Velocity

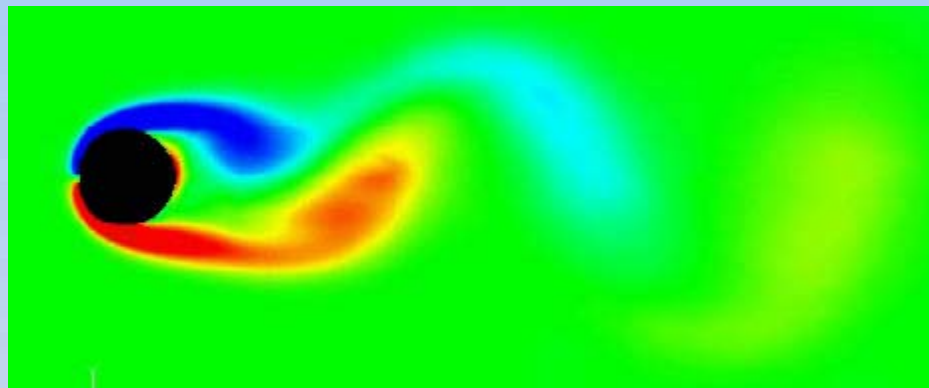


Feedback controlled

Centerline



$\frac{1}{4}$ span





Summary

- Develop feedback flow control strategy based on low dimensional model
 - Global flow state estimation using POD
 - State based feedback controller
- Otherwise, this flow is not controllable



Best use for feedback flow control

- Experiments
 - Initial qualitative flow physics understanding
 - Open-loop parameter scans
 - Final testing of controllers
- Computations
 - Detailed data production of key cases (determined by experiments)
 - Debugging of feedback flow control
 - Data availability, no measurement errors
- Modeling
 - Crucial for controller development
 - Initial controller testing
 - Model provides global flow state estimation for real time implementation



Conclusions

- Integration of theory, experiments and simulations
 - more than the sum of all three
 - evaluate best possible use of each at beginning of project
- Need experts in all involved fields
 - But: each expert needs working knowledge of *all* other fields involved
- Communication is paramount
- *State based feedback flow control impossible without IFD*



Outlook

- Application of developed feedback flow control methodology
 - Higher Reynolds numbers
 - Turbulent flows
- New applications
 - Aero Servo Optics
 - Unsteady aerodynamics (MAV, flapping flight)